

NASA RELIABILITY PREFERRED PRACTICES FOR DESIGN AND TEST

NASA Technical Memorandum 4322A

February 1999

This edition of the NASA Reliability Preferred Practices and Guidelines incorporates the sixth, seventh and eighth supplements to the initial issue of September, 1991. Only Practice #1408 has been updated from the preceding set of 111 practices. The remaining material represents 54 new practices with over 300 pages of heritage design and test information.

A new category of practices for Ground Support Equipment was added in supplement 5 as "Section IV, Ground Support Equipment (GSE) Practices." These practices follow a less complex numbering convention and use only the prefix "GSE" followed by "-3XXX."

A paper copy of the full set of practices is now available, but, in the light of paper reduction, use of the web-based edition is strongly recommended. However, limited copies of the paper version may be obtained from

National Aeronautics and Space Administration
Code QS
300 E Street, SW
Washington, DC 20546

and from your local NASA Center via the Center Contacts listed on page v. Access through the web at <http://www.hq.nasa.gov/office/codeq/overvu.htm> is the most efficient way to access this information.

NASA Technical Memorandum 4322A

NASA Reliability Preferred Practices for Design and Test

NASA Reliability and Maintainability
Steering Committee
NASA Office of Safety and Mission Assurance
Washington, D.C.



National Aeronautics and
Space Administration
Office of Management
Scientific and Technical
Information Program

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6(1996), 7(1997), and 8(1998))

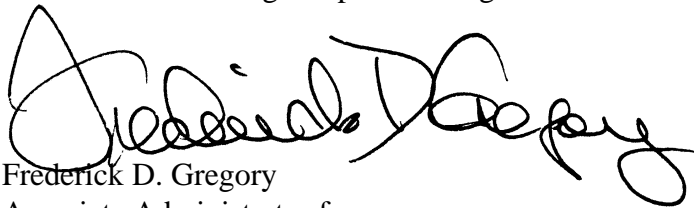
PREFACE

This manual summarizes reliability experience from both NASA and industry. It is intended to reflect engineering principles to support current and future civil space programs.

Reliability must be an integral part of the systems engineering process. The application of sound reliability principles will be the key to an effective and affordable American space program. Experience with our space programs has shown that reliability efforts must focus on the design characteristics that affect frequency of failure. This manual emphasizes that these identified design characteristics must be controlled through the application of conservative engineering principles.

I strongly encourage the use of this manual to assess your current reliability techniques. The manual should promote an active technical interchange between reliability and design engineering that focuses on the potential impact of design margin on maintenance and logistics requirements.

This manual is intended to be a dynamic medium for technical communication. Additional practices and guidelines will be published periodically. This manual should be considered a series of technical memoranda for promoting a systematic approach to the reliability discipline. These practices and guidelines provide the engineering community with useful tools to assure the highest possible degree of success in the Nation's civil space program.

A handwritten signature in black ink, appearing to read 'Frederick D. Gregory', written in a cursive style.

Frederick D. Gregory
Associate Administrator for
Safety and Mission Assurance

NASA Wins IEEE Reliability Society “Company of the Year” Award



NASA and the USAF Rome Laboratory were co-recipients of the Institute of Electrical and Electronics Engineers Reliability Society's first "Company of the Year" Recognition Award at the Society's Annual Awards Banquet in Anaheim, California on January 18, 1998.

In her letter to NASA's Administrator, IEEE Reliability Society President, Loretta Arellano noted,

"NASA's development and placing of the "NASA Reliability Preferred Practices for Design and Test" and the "Recommended Techniques for Effective Maintainability" on the World Wide Web has made available an excellent heritage compendium of aerospace design information to the engineering community. Although the title implies that the information contained in these electronic data sources is the stereotypical "multiple nines" information, I was pleased to find it actually contains engineering design and test information that has lead to the development and launching of the highly successful projects with which our country's Space Agency has distinguished itself and the nation. This information is useful not only to the Product Assurance disciplinarians, but more importantly to the design and test engineers by putting design for mission success information in their language and supplying them with the appropriate tools. “

The Inscription Reads:

THE INSTITUTE OF ELECTRICAL AND ELECTRONICS ENGINEERS, INC.
RELIABILITY SOCIETY

presents the
COMPANY RECOGNITION AWARD
to
NASA

This is to recognize NASA for the "NASA Reliability Preferred Practices for Design and Test" and the "Recommended Techniques for Effective Maintainability." These heritage sources openly share essential success knowledge that might otherwise be lost as the technical workforce changes. This is truly the essence of NASA's "Faster, Better, and Cheaper" goal.

Anaheim, CA
18 January 1998

Loretta Arellano
President, IEEE Reliability Society

CENTER CONTACTS

In the preparation of this manual, the dedication, time, and technical contributions of the following individuals are appreciated. Without the support of their individual centers, and their enthusiastic personal support and willingness to serve on the NASA Reliability and Maintainability Steering Committee, the practices and guidelines contained in this manual would not be possible.

All of the NASA Centers were invited to contribute to this manual. The people listed below may be contacted for more information about these practices and guidelines.

Mr. John W. Remez
NASA Goddard Space Flight Center
Code 302.0 Bldg. 6 Rm. S240
Greenbelt, MD 20771

Mr. James F. Clawson
Jet Propulsion Laboratory
California Institute of Technology
MS 301-456 SEC 505
4800 Oak Grove Drive
Pasadena, CA 91109

Mr. Timothy C. Adams
Lyndon B. Johnson Space Center
National Space and Aeronautics
Administration
Bldg. 45 RM 656 Code NX
2101 NASA Road 1
Houston, TX 77058

Mr. Leon Migdalski
NASA Kennedy Space Center
RT-ENG-2 KSC HQS 3548
Kennedy Space Center, FL 32899

Mr. John A. Greco
NASA Langley Research Center
MS 421, Bldg. 1162A, Room 125
5 Freeman Road
Hampton, VA 23681-0001

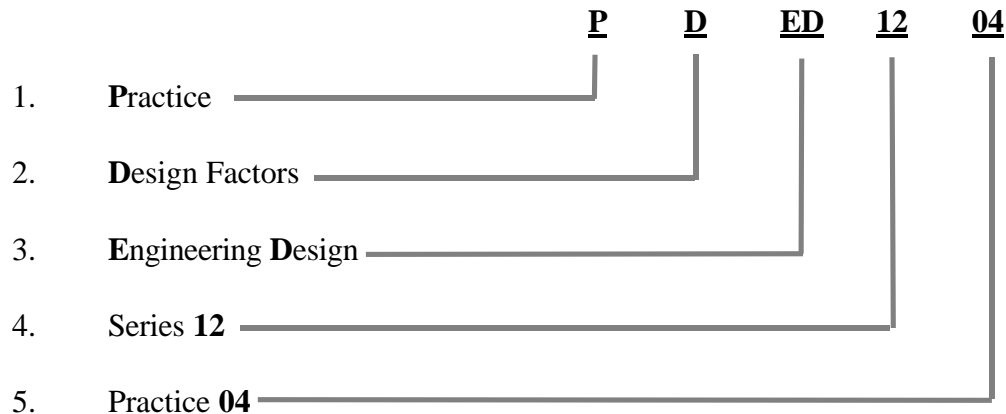
Mr. Vincent R. Lalli
NASA Lewis Research Center
MS 501-4 Code 0152
21000 Brookpark Road
Cleveland, OH 44135

Mr. Frank L. Hepburn
George C. Marshall Space Flight Center
CR10, Bldg. 4203, Room 6428
MSFC Huntsville, AL 35812

Mr. Wilson B. Harkins, III
NASA Headquarters Code QS
300 E Street, SW
Washington, DC 20546

DOCUMENT REFERENCING

The following shows the document numbering system¹ applicable to these practices and guidelines. The example illustrated here is “Part Junction Temperature,” Practice No. PD-ED-1204.



Key to Nomenclature

<u>Position</u>	<u>Code</u>
1.	G - Guideline P - Practice
2.	D. - Design Factors T - Test Elements
3.	EC - Environmental Considerations ED - Engineering Design AP - Analytical Procedures TE - Test Considerations & Procedures
4.	x - Series Number
5.	xx - Practice Number within Series

¹A separate category of practices for Ground Support Equipment was added in supplement #5. These practices follow a less complex numbering convention and use only the prefix “GSE” followed by “-3XXX.”

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1100 Series --Environmental Considerations

- **1101 - Environmental Factors**
- **1102 - Meteoroids/Space Debris**
- **1103 - Nickel-Cadmium Conventional Spacecraft Battery Handling and Storage Practice**
- **1104 - Monitoring DC Magnetic Field Peak Exposure**
- **1105 - Solar Flare Proton and Heavy Ion Modeling for Single Event Effects**
- **1106 - Plasma Noise in EMI Design**

1200 Series -- Engineering Design

- **1201 - EEE Parts Derating**
- **1202 - High Voltage Power Supply Design and Manufacturing Practices**
- **1203 - Class S Parts in High Reliability Applications**
- **1204 - Part Junction Temperature**
- **1205 - Welding Practices for 2219 Aluminum and Inconel 718**
- **1206 - Power Line Filters**
- **1207 - Magnetic Design Control for Science Instruments**
- **1208 - Static Cryogenic Seals for Launch Vehicle Applications**
- **1209 - Ammonia-Charged Aluminum Heat Pipes with Extruded Wicks**
- **1210 - Assessment and Control of Electrical Charges**
- **1211 - Combination Methods for Deriving Structural Design Loads Considering Vibro-Acoustic, etc., Responses**
- **1212 - Design and Analysis of Electronic Circuits for Worst Case Environments and Part Variations**

- 1213 - Electrical Shielding of Power, Signal, and Control Cables
- 1214 - Electrical Grounding Practices for Aerospace Hardware
- 1215.1 Preliminary Design Review
- 1215.2 Hardware Review/Certification Requirement
- 1215.3 Critical Design Review for Unmanned Missions
- 1215.4 Common Review Methods for Engineering Products
- 1215.5 Pre-Ship Review
- 1216 - Active Redundancy
- 1217 - Structural Laminate Composites for Space Applications
- 1218 - Application of Ablative Composites to Nozzles for Reusable Solid Rocket Motors
- 1219 - Vehicle Integration/Tolerance Build-up Practices
- 1220 - Demagnetization of Ferromagnetic Parts
- 1221 - Battery Selection Practice for Aerospace Power Systems
- 1222 - Magnetic Field Restraints for Spacecraft Systems and Subsystems
- 1223 - Vacuum Seals Design Criteria
- 1224 - Design Considerations for Fluid Tubing Systems
- 1225 - Conducted and Radiated Emissions Design Requirements
- 1226 - Thermal Design Practices for Electronic Assemblies
- 1227 - Controlling Stress Corrosion Cracking in Aerospace Applications
- 1228 - Independent Verification and Validation of Embedded Software
- 1229 - Selection of Electric Motors for Aerospace Applications
- 1230 - System Design Analysis Applied to Launch Vehicle Configuration
- 1231 - Design Considerations for Lightning Strike Survivability
- 1232 - Spacecraft Orbital Anomaly Report (SOAR) Systems
- 1233 - Contamination Control Program
- 1234 - GPS Timing System
- 1235 - Over-Speed Protection System for DC Motor Driven Cranes
- 1236 - EEE Parts Selection Guidelines for Flight Systems
- 1237 - (Practice Number reserved for future supplement)
- 1238 - Spacecraft Electrical Harness Design Practice
- 1239 - Spacecraft Thermal Control Coatings Design and Application Procedures
- 1240 - Guidelines for the Identification, Control, and Management of Critical Items
- 1241 - Contamination Budgeting of Space Optical Systems
- 1242 - Design Considerations for Space Trusses
- 1243 - Fault Protection
- 1244 - Design Practice to Control Interface from Electrostatic Discharge (ESD)

- 1245 - Magnetic Dipole Allocation
- 1246 - Fault Tolerant Design
- 1247 - Spacecraft Lessons Learned Reporting System
- 1248 - Spacecraft Data Systems (SDS) Hardware Design Practices
- 1249 - Electrostatic Discharge (ESD) Control in Flight Hardware
- 1250 - Pre-Flight Problem / Failure Reporting Procedures
- 1251 - Instrumentation System Design and Installation for Launch Vehicles
- 1252 - Material Selection Practices
- 1253 - Arcjet Thruster Design Considerations for Satellites
- 1254 - Design Reliable Ceramic Components with CARES Code
- 1255 - Problem Reporting and Corrective Action System
- 1256 - Automatic Transfer Switch (ATS) in Critical Applications
- 1257 - Solid Rocket Motor Joint Reliability
- 1258 - Space Radiation Effects on Electronic Components in Low-Earth Orbit
- 1259 - Acoustic Noise Requirements
- 1260 - Radiation Design Margin Requirement
- 1261 - Characterization of RF Subsystem Susceptibility to Spurious Signals
- 1262 - Subsystem Inheritance Review
- 1263 - Contamination Control of Space Optical Systems
- 1264 - Integrated Optical Performance Modeling of X-Ray Systems
- 1265 - Precision Diamond Turning of Aerospace Optical Systems
- 1266 - Binary and Hybrid Optics for Space Applications
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1300 Series -- Analytical Procedures

- 1301 - Surface Charging and Electrostatic Discharge Analysis
- 1302 - Independent Review of Reliability Analyses
- 1303 - Part Electrical Stress Analysis
- 1304 - Problem/Failure Report Independent Review/Approval
- 1305 - Risk Rating of Problem/Failure Reports
- 1306 - Thermal Analysis of Electronic Assemblies to the Piece Part Level
- 1307 - Failure Modes, Effects, and Criticality Analysis (FMECA)
- 1308 - Electromagnetic Interference Analysis of Circuit Transients
- 1309 - Analysis of Radiated EMI from ESD Events Caused by Space Charging
- 1310 - Spurious Radiated Interference Awareness
- 1311 - Computational Fluid Dynamics (CFD) in Launch Vehicle Applications
- 1312 - The Team Approach to Fault-Tree Analysis

- 1313 - System Reliability Assessment Using Block Diagramming Methods
- 1314 - Sneak Circuit Analysis for Electromechanical Systems
- 1315 - Redundancy Switching Analysis
- 1316 - Thick Dielectric Charging / Internal Electrostatic Discharge (IESD)
- 1317 - Flight Load Analysis as a Spacecraft Design Tool
- 1318 - Structural Stress Analysis

1400 Series -- Test Elements

- 1401 - EEE Parts Screening
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- 1403 - Thermographic Mapping of PC Boards
- 1404 - Thermal Test Levels
- 1405 - Powered-On Vibration
- 1406 - Sinusoidal Vibration
- 1407 - Assembly Acoustic Tests
- 1408 - Pyrotechnic Shock
- 1409 - Thermal Vacuum Versus Thermal Atmospheric Test of Electronic Assemblies
- 1410 - Selection of Spacecraft Materials and Supporting Vacuum Outgassing Data
- 1411 - Heat Sinks for Parts Operated in Vacuum
- 1412 - Environmental Test Sequencing
- 1413 - Random Vibration Testing
- 1414 - Electrostatic Discharge (ESD) Test Practices
- 1415 - Power System Corona Testing
- 1416 - Radiated Susceptibility System Verification
- 1417 - Electrical Isolation Verification (DC)
- 1418 - Qualification of Non-Standard EEE Parts in Spaceflight Applications
- 1419 - Vibroacoustic Qualification Testing of Payloads, Subsystems, and Components
- 1420 - Sine-Burst Load Test
- 1421 - Eddy Current Testing of Aerospace Materials
- 1422 - Ultrasonic Testing of Aerospace Materials
- 1423 - Radiographic Testing of Aerospace Materials
- 1424 - Leak Testing of Liquid Hydrogen and Liquid Oxygen Propellant Systems
- 1425 - Magnetic Particle Testing of Aerospace Materials
- 1426 - Penetrant Testing of Aerospace Materials
- 1427 - Rocket Engine Technology Test Bed Practice
- 1428 - Practice of Reporting Parts, Materials, and Safety Problems (Alerts)

- 1429 - Integration & Test Practices to Eliminate Damage to Flight Hardware
- 1430 - Short Circuit Testing for Nickel Hydrogen Battery Cells
- 1431 - Voltage/Temperature Margin Testing
- 1432 - RF Breakdown Characterization
- 1433 - Mechanical Fastener Inspection System
- 1434 - Battery Verification through Long Term Simulation
- 1435 - Verification of RF Hardware Design Performance
- 1436 - Advanced Computed X-Ray Tomography
- 1437 - End-to-End Compatibility and Mission Simulation Testing (TBS)
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- 1439 - Systems Test Considerations for High Performance Liquid Propellant Rocket Engines

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- 2200 Series -- Engineering Design
- 2201 - Fastener Standardization and Selection Considerations
- 2202 - Design Considerations for Selection of Thick-Film Microelectronic Circuits
- 2203 - Design Checklists for Microcircuits
- 2204 - Concurrent Engineering Guideline for Aerospace Systems
- 2205 - Design and Manufacturing Guideline for Aerospace Composites
- 2206 - Selection of Compatible Materials for Use with Fluorine
- 2207 - Designing for Dormant Reliability
- 2208 - Fabrication of Gaseous and Liquid Fluorine Systems
- 2209 - Spacecraft Deployed Appendage Design Guidelines
- 2210 - Fiber-Reinforced Polymer Composite Material Selection
- 2211 - Coordinate Systems for Attitude Determination and Control
- 2213 - Management of Limited Failure Analysis Resources for EEE Parts
- 2214 - Marman Clamp System Design Guidelines

2300 Series -- Analytical Procedures

- 2301 - Earth Orbit Environmental Heating
- 2302 - Thermal Analysis of Spacecraft Hardware Guideline

- 2303 - Spectral Fatigue Reliability
- 2304 - Fracture Mechanics Reliability
- 2305 - Structural Analysis in the Design of Optical Mirrors

2400 Series -- Test Elements

- 2401 - EMC Guideline for Payloads, Subsystems, and Components
- 2402 - Near Field Measurement for Large Aperture Antenna Pattern Determination
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- 3002 - Fail Safe Firex/Deluge System
- 3003 - Redundancy in Critical Mechanical Systems
- 3004 - Use of Design Review Checklists for Space Shuttle GSE
- 3005 - Identification, Control, and Management of GSE Critical Items
- 3006 - Environmental Test Methods for Ground Support Equipment
- 3007 - Redundancy Considerations for Ground Communication Systems
- 3008 - Electrostatic Discharge Control for GSE
- 3009 - Uninterruptable Power Supply Systems (UPS)
- 3010 - Oil-Free Vacuum Pump in the LOX/LH2 Transfer System
- 3011 - Foreign Object Debris (FOD) Programs at KSC

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OVERVIEW

A. PURPOSE

This manual is produced to communicate within the aerospace community design practices that have contributed to NASA mission success. The information presented has been collected from various NASA field centers and reviewed by a committee consisting of senior technical representatives from the participating centers.

B. APPLICABILITY

The information presented in this manual represents the “best technical advice” that NASA has to offer on reliability design and test practices. The practices contained in this manual should not be interpreted as requirements, but rather as proven technical approaches that can enhance system reliability. Application of the practices and guidelines is strongly encouraged, but the final decisions regarding applicability resides with the particular program or project office

The manual is divided into three technical sections. Part I contains reliability practices, including design criteria, test procedures, and analytical techniques that have been successfully applied on previous space flight programs. Part II contains reliability guidelines, including techniques currently applied to space flight projects where insufficient information exists to certify that the technique will contribute to mission success. Part III contains design, test, and procedural practices that have contributed to successful ground support of space flight and ground-based aerospace programs.

C. DISCUSSION

Experience with NASA’s successful extended duration space missions shows that four fundamental elements contribute to high reliability:

- 1) Understanding stress factors imposed on flight hardware by its operating environment;
- 2) Controlling the stress factors through selection of conservative design criteria;
- 3) Appropriate analysis to identify and track high stress points in the design (prior to qualification testing or flight use); and
- 4) Careful selection of redundancy alternatives that will provide the necessary function(s) should failure occur.

This manual is provided to encourage design, test, and reliability engineers to give careful attention to both redundancy and management of stress factors during the design and development of space flight systems.

D. CONTROL/CONTRIBUTION

The practices and guidelines contained in this manual serve as a mechanism for communicating the latest techniques that contribute to high reliability. This publication will be revised periodically to include additional practices/guidelines, or revisions to information (as additional technical data becomes available). Contributions from aerospace contractors and NASA Field centers is encouraged. Any practice, guideline or technique that appears appropriate for inclusion in this manual should be submitted for review. Submissions should be formatted identically to the practices and guidelines in this manual and sent to the address below for consideration:

National Aeronautics and Space Administration
Code QS
300 E Street, SW
Washington, DC 20546

Organizations submitting practices/guidelines that are selected for inclusion in this manual will be recognized in the lower right-hand corner of the published item.

PART I: RELIABILITY PRACTICES

A. INTRODUCTION

This section contains Reliability Design Practices that have contributed to the success of previous space flight programs. The information presented in this section is for use throughout NASA and the aerospace community to assist in the design and development of highly reliable equipment and assemblies. The practices include recommended analysis procedures, redundancy considerations, parts selection, environmental requirements considerations, and test requirements and procedures.

B. RELIABILITY DESIGN PRACTICE FORMAT DEFINITIONS

The format for the reliability practices is shown below.

PRACTICE FORMAT DEFINITIONS

Practice: *A brief statement of the practice*

Benefit: *A concise statement of the technical improvement realized from implementing the practice*

Programs That Certified Usage: *Identifiable programs or projects that have applied the practice*

Center to Contact for More Information: *Source of additional information, usually the sponsoring NASA Center. See "CENTER CONTACTS", page v*

Implementation Method: *A brief technical discussion that is not intended to give the full details of the process, but rather to provide a design engineer with adequate information to understand how the practice should be used.*

Technical Rationale: *A brief technical justification for the use of the practice*

Impact of Nonpractice: *A brief statement of what can be expected if use of the practice is avoided*

Related Practices: *Identification of other topic areas in the manual that contain related information*

References: *Publications that contain additional information about the practice*

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OF
PRACTICE**

PART II: RELIABILITY DESIGN GUIDELINES

A. INTRODUCTION

This section contains Reliability Design Guidelines for consideration by the aerospace community. The guidelines presented in this section contain valuable information that in the opinion of the sponsoring activity, represents a technically credible process that could be applied to ongoing NASA programs/projects. Unlike a Reliability Design Practice, a guideline lacks specific operational experience or data to indicate that a topic area has contributed to mission success. However, a guideline does contain information that represents current “best thinking” on a particular topic and is a well thought out approach to resolving a particular issue or problem. There is a unanimous Reliability and Maintainability Steering Committee agreement with the appropriateness of application of the approach.

B. RELIABILITY GUIDELINE FORMAT DEFINITIONS

The format for the reliability guidelines is shown below

GUIDELINE FORMAT DEFINITIONS

Guideline: *A brief statement of the guideline*

Benefit: *A concise statement of the technical improvement realized from implementing the guideline*

Center to Contact for More Information: *Source of additional information, usually the sponsoring NASA Center. See “CENTER CONTACTS”, page iii*

Implementation Method: *A brief technical discussion that is not intended to give the full details of the process, but rather to provide a design engineer with adequate information to understand how the guideline should be used.*

Technical Rationale: *A brief technical justification for the use of the guideline*

Impact of Nonpractice: *A brief statement of what can be expected if use of the guideline is avoided*

Related Practices: *Identification of other topic areas in the manual that contain related information*

References: *Publications that contain additional information about the guideline*

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PART III: GROUND SUPPORT EQUIPMENT (GSE) PRACTICES

A. INTRODUCTION

This section contains design and procedural practices that have contributed to successful ground support of space flight and ground-based aerospace programs. The information presented in this section is for use throughout NASA and the aerospace community to assist in the design, development, and operation of highly reliable ground support equipment and assemblies. This material is primarily concerned with design and test techniques, procedures for control of critical items, and control of environmental influences on successful launch.

B. RELIABILITY DESIGN PRACTICE FORMAT DEFINITIONS

The format for the reliability practices is shown below.

PRACTICE FORMAT DEFINITIONS

Practice: *A brief statement of the practice*

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